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A SIMULATION-DRIVEN DESIGN APPROACH**

Lennart Karlsson, Andreas Pahkamaa, Magnus Karlberg, Magnus Lofstrand, John Goldak  
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## MECHANICS OF MATERIALS AND STRUCTURES: A SIMULATION-DRIVEN DESIGN APPROACH

LENNART KARLSSON, ANDREAS PAHKAMAA, MAGNUS KARLBERG,  
MAGNUS LÖFSTRAND, JOHN GOLDAK AND JONAS PAVASSON

Engineering product development has developed considerably over the past decade. In order for industry to keep up with continuously changing requirements, it is necessary to develop new and innovative simulation methods. However, few tools and methods for simulation-driven design have been applied in industrial settings and proven to actually drive the development and selection of the ideal solution. Such tools, based on fundamental equations, are the focus of this paper.

In this paper the work is based on two cases of mechanics of materials and structures: welding and rotor dynamical simulations. These two examples of simulation-driven design indicate that a larger design space can be explored and that more possible solutions can be evaluated. Therefore, the approach improves the probability of innovations and finding optimal solutions.

A calibrated block dumping approach can be used to increase the efficiency of welding simulations when many simulations are required.

### 1. Introduction

Engineering product development has evolved considerably over the past decade. The vision of the sustainable society and a global context for many industries is now forcing manufacturing companies to develop innovative ways of developing their products. In addition, the product life-cycle will likely be shorter in the future, partly because customers will demand shorter delivery times. Suppliers will have to be more efficient than today and be able to assemble products from a large number of parts from subsuppliers. Increasingly, the value of the product originates outside the physical artifact (the hardware), in the form of services and add-ons. Instead of hardware ownership, other needs such as availability, productivity and risk minimization increasingly need to be catered for. In such a challenging environment and in order for industry to keep up with demand, it is necessary to develop new and innovative simulation methods, knowledge management methods and tools and methods for distributed work, to name a few.

In this paper, the focus is mainly on simulation-driven design applied on welding simulations (mechanics of materials) and rotor dynamical design (mechanics of structures). Below, the basic equations for modeling and simulation of welding and the basic equations for modeling and simulation of nonlinear dynamics are solved. For these two important problems, the way in which this fundamental modeling can effectively be used in simulation-driven product development is demonstrated.

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*Keywords:* simulation-driven design, calibrated block dumping, welding simulations, rotor dynamical simulations.



**Figure 1.** An example of a product development process.

Most industrial companies of any size follow some development process when developing their products. [Figure 1](#) shows an example of such a development process from planning to production ramp-up [[Ulrich and Eppinger 1995](#)].

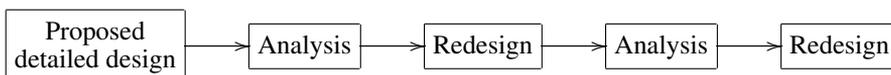
Product development literature provides a broad view of how to understand customer needs, develop and sell products, and includes discussions concerning best practices [[Ulrich and Eppinger 1995](#); [Wheelwright and Clark 1992](#); [Cross 2000](#)]. For example, [Smith and Reinertsen \[1998\]](#) offer a general view and aim to describe methods for generating a product to meet customer needs. Recently, product development processes have been extended to cover the whole process from needs to recycling.

Within the hardware product development domain, numerous tools have been developed to support the creation of excellent goods; e.g., computer aided engineering for geometric representation [[LaCourse 1995](#)] and the finite element method for stress calculation, exemplified in this paper. Typically, this work has been about making knowledge explicit and expressible and support tools have over time been developed to aid the creation of the hardware.

Simulations are ideally supposed to be used in the earliest stages of the development process. Often, however, there are few analyses involved until the detailed design stage is reached. At this stage simulations are often used in a verifying sense, meaning that the designer submits a model to the analysts who perform the analysis and then deliver the results to the designer. The results are used by the designer to verify whether or not the proposed design will meet the criteria. If not, the designer needs to upgrade the initial design, and then propose a new design for the analysts who perform a new simulation and so on, see [Figure 2](#).

This sequential simulation usage workflow in combination with little usage of simulations in the earliest stages of the development process gives an ineffective product development process. To overcome this problem strategies for simulation-driven development have been proposed by some researchers [[Courter 2009](#); [Sellgren 1995](#); [Wall 2007](#)]. Naturally, such work has been ongoing for a relatively long time, as exemplified by [[Hansen 1974](#)] and [[Gero 1981](#)]. The development processes discussed in literature are often analyzed from an abstract point of view, while few reports exist on how simulation tools shall be designed and used to enable such methodologies. That is, few tools and methods have been applied in industrial settings and proven to actually drive the development and selection of the ideal solution.

Given these challenges, this paper is focused on briefly introducing the area of simulation driven design (SDD) [[Courter 2009](#)], and based on that approach describing a block dumping technique applied in welding simulations. Another application of the SDD approach is presented in the application area of rotor dynamical design.



**Figure 2.** Sequential verifying simulation usage workflow.

Thus, the objective of this paper is to show how tools for modeling and simulation of mechanics of materials and structures shall be designed and used to enable simulation-driven product development. The objective is exemplified through two cases of mechanics of materials and structures: welding and rotor dynamical simulations.

## 2. Simulation-driven design methodology

In this paper a simulation-driven methodology is used [Bylund 2004; Goldak et al. 2007]. Bylund states that one way of reaching simulation-driven design is by providing simulation tools that can be used by the designers themselves. Designers thereby avoid having to send models and results back and forth, often among several different people. For designers not used to analysis work to be able to perform the necessary analyses, tools that are *intuitive and include expert knowledge* through equations of mechanics of materials and structures are required. The results from the analyses must further be *derived quickly* and must be *easy to interpret*. Hence, the demands on the simulation tools affect the preprocessing, the solver and the postprocessing.

## 3. Simulation-driven design of welded structures

Welding is one of the most commonly used methods of joining metal pieces. In fusion welding, the metal pieces are heated until they melt together, resulting in strong coupling between thermal, mechanical and metallurgical (microstructural) properties. Due to the complexity of the welding applications, the governing equations as well as the thermal, mechanical and metallurgical couplings, computational support is necessary for prediction of distortions and residual stresses. The development of the field of computational welding mechanics has been described in, for example, [Karlsson 1986; Goldak and Akhlaghi 2005; Lindgren 2001a; 2001b; 2001c; 2007]. Much research has focused on predictions of residual stresses and distortions [Chen and Sheng 1992; Lee et al. 2008; Ueda et al. 1988; Ueda and Yuan 1993].

Several reports exist regarding validation of predictions of residual stresses and deformation [Barroso et al. 2010; Karlsson et al. 1989; Mochizuki et al. 2000; Ueda et al. 1986]. The European Network on Neutron Techniques Standardization for Structural Integrity (NeT) has a benchmark problem for a single weld bead-on-plate specimen [Truman and Smith 2009]. By using finite element simulations, NeT members have predicted and measured residual stresses and thermal fields of this benchmark problem by different methods, the results of which have been compiled [Smith and Smith 2009a; 2009b]. In those reports, different sources of errors are discussed, and it was also found that “there is much room for improvement” regarding prediction accuracy.

In product development, residual stresses and distortions often need predicting to verify alignment tolerances, strength demands, fatigue requirements, etc. It is important, for example, to keep track of the residual stress- and deformation history when simulating a sequence of manufacturing processes [Deng et al. 2009; Åström 2004]. Different approaches have been developed for how to use welding simulations to predict suitable sequences of weld paths. Troive et al. [1998] compared different predefined paths, while Voutchkov et al. [2005] used surrogate models to solve a combinatorial weld path planning problem. Efforts have further been made to show effects of deposition sequences regarding, for instance, residual stresses and distortions [Ogawa et al. 2009]. Although a lot of research has been conducted on welding

simulations, there is still a need for development of more efficient welding simulations and methodologies for using such simulations to support design processes.

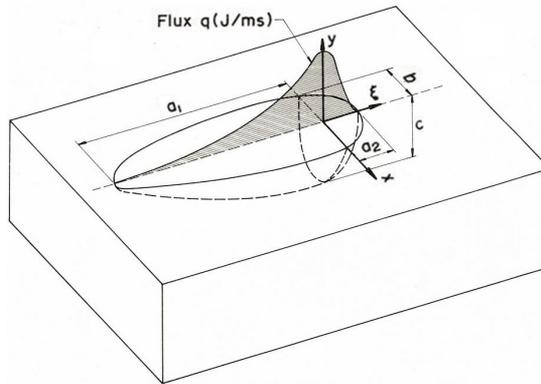
In this article, a strategy for improved efficiency of welding simulations is used as proposed in [Pahkamaa et al. 2010], aiming at developing a simulation-driven design methodology for welding simulations. In this work, the modeling and simulation software VrWeld from Goldak Technologies Inc. [Goldak Technologies 2010] has been used to demonstrate the proposed strategy.

**3.1. *Welding simulation theory.*** The field of welding simulations described in this paper (computational welding mechanics) is partially built upon earlier work within the fields of thermal, mechanical and metallurgical (microstructural) properties of materials. The principles and applications of welding simulations have been described in [Goldak and Akhlaghi 2005; Lindgren 2001b]. Welding simulation is a good example of how mechanics of materials and structures can be put to practical use with the support of computers. The essential features of computational weld mechanics (CWM) are [Goldak 2009]:

- It requires solving the nonlinear, coupled 3D transient partial differential equations (PDEs) for heat flow (conservation of energy), microstructure evolution and stress-strain evolution (conservation of momentum).
- The material properties are temperature, history dependent and involve phase changes.
- The welding process usually adds material (filler metal), making the geometric domain a time-dependent free surface problem.
- The boundary conditions applied by fixtures, clamps and tack welds are complex and transient.
- The geometry of welded structures is often complex with many parts.
- Modeling the heat source of the arc is itself complex.

**3.1.1. *Welding simulation software (VrWeld).*** VrWeld, part of the VrSuite software package, is a finite element program used to simulate welding processes. VrSuite [Goldak Technologies 2010] uses operator splitting to solve the system of equations needed to model manufacturing processes such as welding and heat treating and the in-service behavior of assemblies of such manufactured parts. Each equation in the system is solved with the algorithm, domain, parameters, initial conditions, boundary conditions, length scale and time scale that best approximate the physics of the phenomena that the equation models. Maps are created to map data from this equation to each equation that it is coupled to. Each equation is solved iteratively using solvers such as frontal, multifrontal, ICCG, GMRES, MG and AMG [Saad 1996]. Limits and bifurcation points are computed using the Wriggers–Simo algorithm [Wriggers and Simo 1990]. A version of Crisfield’s arc-length method [Ramm 1981] is used to follow the path of the solution in nonlinear analysis. The meshing is largely automated. Domain decomposition and adaptive meshing play an important role. Solvers run in-core using processors with 8 GB of RAM. The CPU-intensive code is written in C++. The high-level code is written in a scripting language such as Tcl/Tk or HTML.

**3.1.2. *Block dumping (fast simulations in VrWeld).*** Traditional transient welding simulations often use a moving heat source such as the Gaussian double ellipsoid [Goldak and Akhlaghi 2005]; see Figure 3. This approach has proved to accurately predict the temperature history, residual stresses and welding distortions [Goldak and Akhlaghi 2005; Lindgren 2001a; 2001b; 2001c; 2007]. It is suggested that the

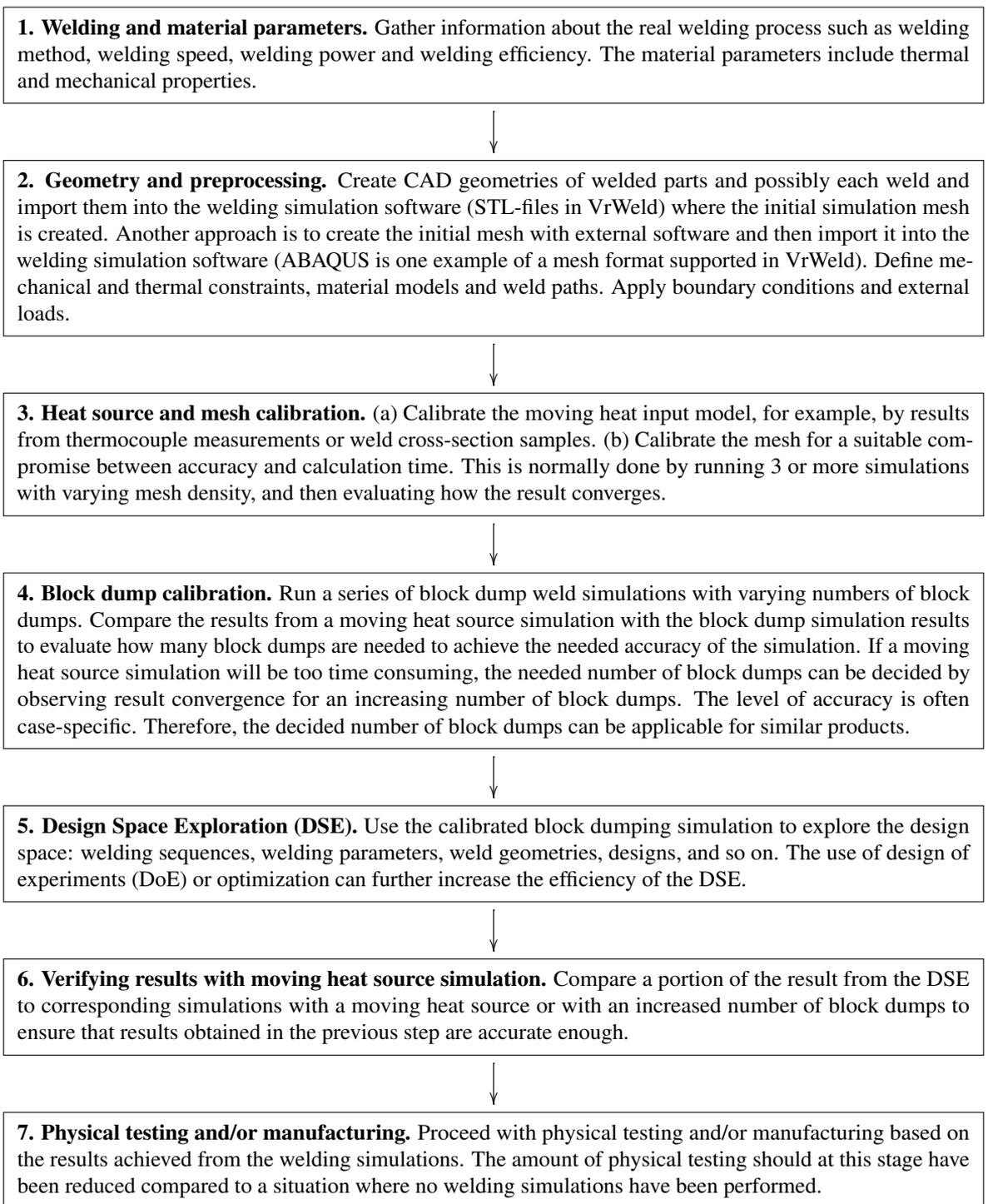


**Figure 3.** Double ellipsoid heat source with Gaussian heat distribution [Goldak and Akhlaghi 2005].

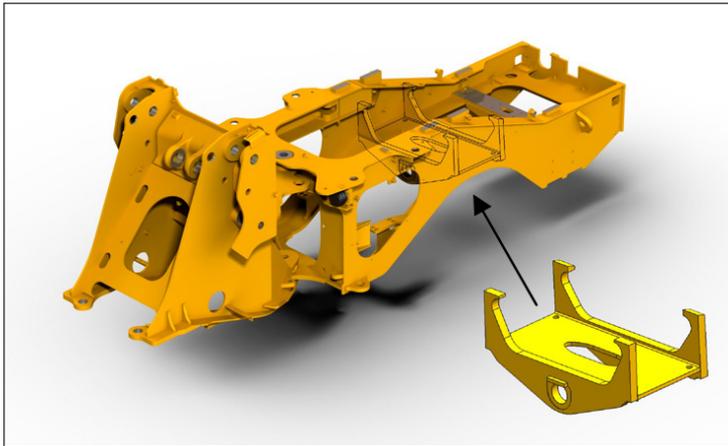
heat source may move no more than half of the weld pool length to function properly in three dimensional welding simulations [Goldak et al. 1986]. This suggestion causes a transient welding simulation that uses a moving heat source to be quite time consuming, even with present-day computational power. The problem gets even more difficult when a large design space is to be explored, meaning that many simulations are required.

A faster way to simulate the thermal history of a weld is the block dumping method [Goldak et al. 1986], also referred to as the prolonged Gaussian heat source [Cai and Zhao 2003]. The block dumping approach heats the whole weld, or large pieces of it, in a single time step. This approach can reduce the number of time steps in the welding simulation significantly and thereby reduce the total calculation time without any appreciable decrease in accuracy [Pahkamaa et al. 2010]. With the block dumping technique, the same parameters as the Gaussian Ellipsoid heat source are used to calculate the amount of heat distributed to each element, i.e., voltage, current, efficiency, welding speed and parameters  $a_1$ ,  $a_2$ ,  $b$  and  $c$  in Figure 3. The total amount of heat distributed to the model is identical to the heat input of a traditional transient simulation. The difference is the time in which the heat is diffused. In each block dump, the amount of heat added is exactly the same as the heat that would be added in a transient analysis for this length of weld, and it is added in exactly the same way with the same transient time steps. However, no heat diffusion is done until all heat has been added for this length of weld. Then one heating time step is done with these applied nodal thermal loads to heat this length of weld. Following this heating time step, the heat equation is solved for the structure as it cools with exponentially increasing length of time steps. Usually, 4 to 6 cooling time steps are applied. The designer can choose to allow each block dump to cool to ambient temperature or use a shorter cooling time in which the structure does not fully cool, the latter approach is used in this paper. In the heating and cooling steps, the thermal-elastoplastic stress analysis problem is solved. This process is applied for each block dump in the sequence of block dumps specified by the designer.

**3.2. Simulation strategy.** The welding simulation strategy proposed in [Pahkamaa et al. 2010] (Figure 4), which will be verified in this study, reduces the calculation time for welding simulations by replacing the traditional moving heat source with a calibrated block dumping heat source. The block dumping heat



**Figure 4.** Suggested welding simulation strategy for simulation-driven design. From [Pahkamaa et al. 2010].



**Figure 5.** Rear frame and rear axle bridge.

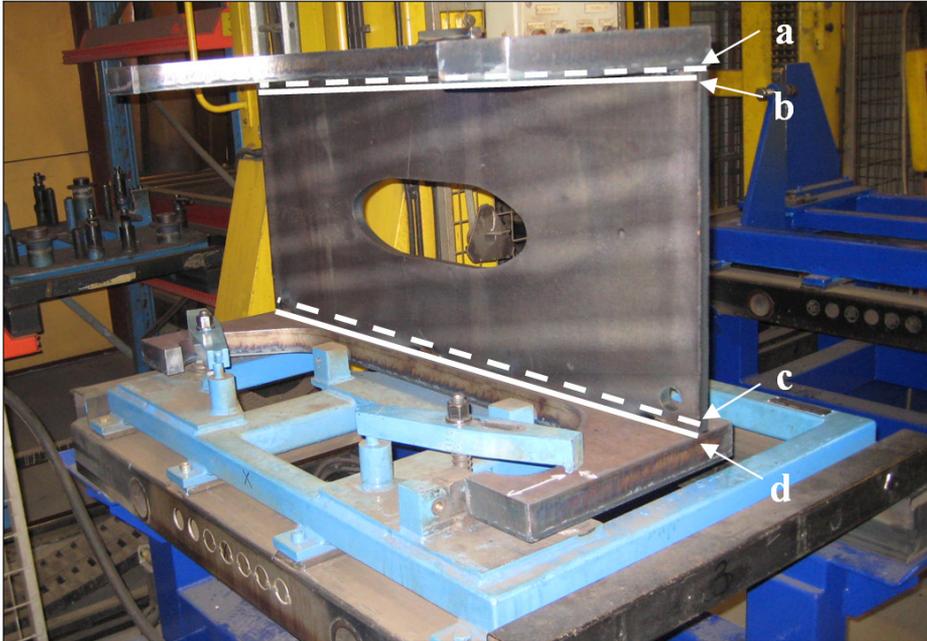
source is calibrated to give a good compromise between simulation accuracy and calculation time, thus allowing a larger design space to be explored during a given period of time. A rear axle bridge from a Volvo Construction Equipment wheel loader is used as a demonstrator case to show that steps 1–6 work on an industrial application. Twenty different welding sequences are compared to find the welding sequence that gives the smallest welding distortion at key positions.

**3.3. Welding case study.** The object of this case study is a rear axle bridge from a Volvo Construction Equipment wheel loader. The axle bridge is positioned in the rear frame as in [Figure 5](#). An axle working as a pivot for the rear axle assembly is mounted in the two holes in the axle bridge. The axle is supported by journal bearings, which are lubricated with oil from the rear axle differential. The concentricity between the two holes and the parallelism and perpendicular alignment between the plates are two important tolerance demands for the axle bridge. Due to these requirements, it is important that the welding process used to manufacture the axle bridge does not introduce excessive deformations. Therefore, the aim of this case study is to derive a suitable welding sequence that minimizes the welding distortion in the axle bridge. Identifying the proper welding sequences is done by use of the proposed simulation strategy; see [Figure 4](#).

**3.3.1. Welding and material properties.** The three plates are joined by welds, marked a–d in [Figure 6](#); the welds have a throat size of 6 mm. The axle bridge is manually tack-welded with 40 mm long tack welds at start, mid and end of the four welds. The tack welding is performed in a separate fixture; the tack welded axle bridge is then positioned in the welding fixture shown in [Figure 6](#). An automated MAG welding process then applies the four welds, with the following parameters:

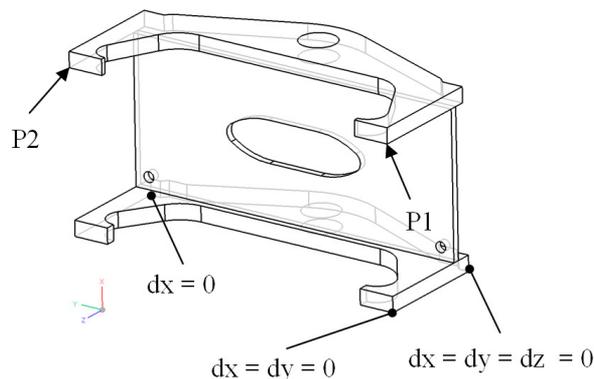
Power : 10880 W (34 V, 320 A)  
Efficiency : 85%  
Welding speed : 37 cm/min

**3.3.2. Geometry and preprocessing.** A CAD model of the axle bridge assembly was provided by the manufacturer. This CAD model was imported to Siemens PLM NX6, where small features were removed to ease the meshing process. The idealized part shown in [Figure 7](#) was exported to the welding

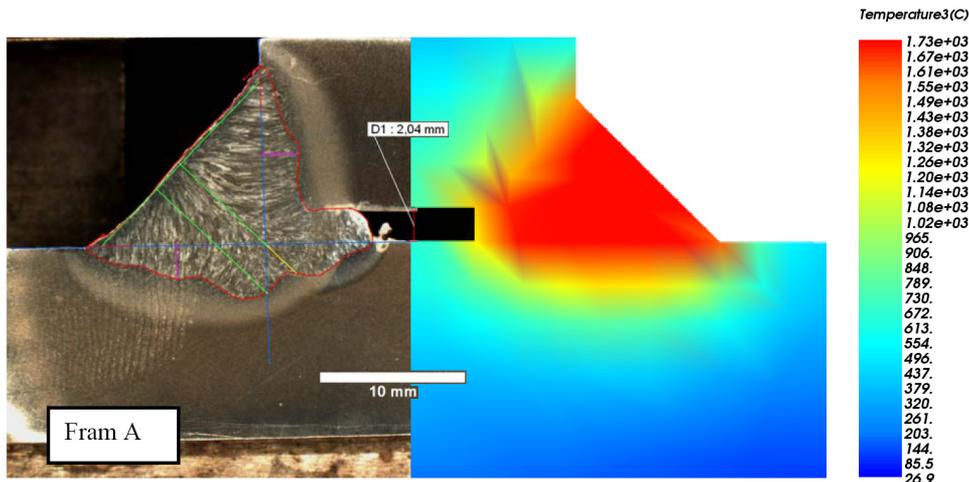


**Figure 6.** Real axle bridge placed in its welding fixture. The weld joints (a–d) are marked with white lines; welds a and c are placed underneath the center plate.

simulation software (VrWeld) in STL-format. The simulation model was then created in VrWeld by defining weld joints, assigning materials, creating an initial mesh and assigning initial conditions and boundary conditions, etc. The material model described in [Andersson 1978] was used for both plates and filler metal. The used fixture shown in Figure 7 prevents rigid body motions (locks six DOFs) without restraining the growth/shrinkage of the plates. This fixture modeling method was used, since the real fixture only clamps one of the plates, and should therefore not have a large impact on the welding distortions. The welding distortion in the rear axle bridge was measured as the global displacement of points P1 and P2. The global ambient temperature is set to 300° K.



**Figure 7.** Idealized axle bridge, evaluation points and constraints.



**Figure 8.** Comparison of weld cross section photograph and simulated heat input.

**3.3.3. Moving heat input calibration.** The moving heat input model was calibrated by use of a weld cross section photograph, see [Figure 8](#). The trial and error method was used to find suitable heat input parameters that describe the shape of the Double Ellipsoid heat source. The calibrated parameters are as follows (see [Figure 3](#)):

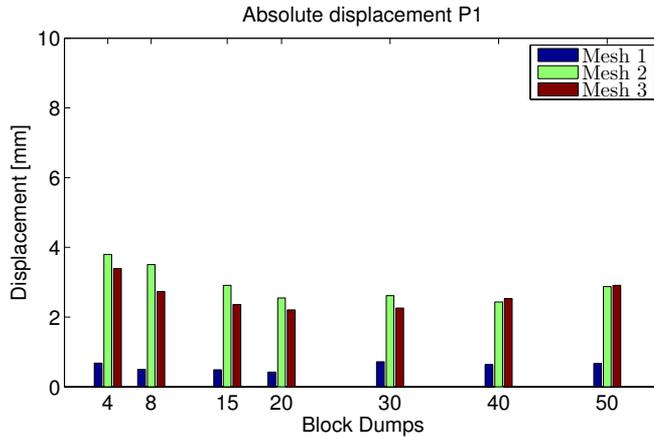
$$a_1 = 10 \text{ mm} \quad a_2 = 6 \text{ mm} \quad b = 6 \text{ mm} \quad c = 8 \text{ mm}$$

**3.3.4. Mesh and block dump calibration.** Three models with different mesh density were created, as follows:

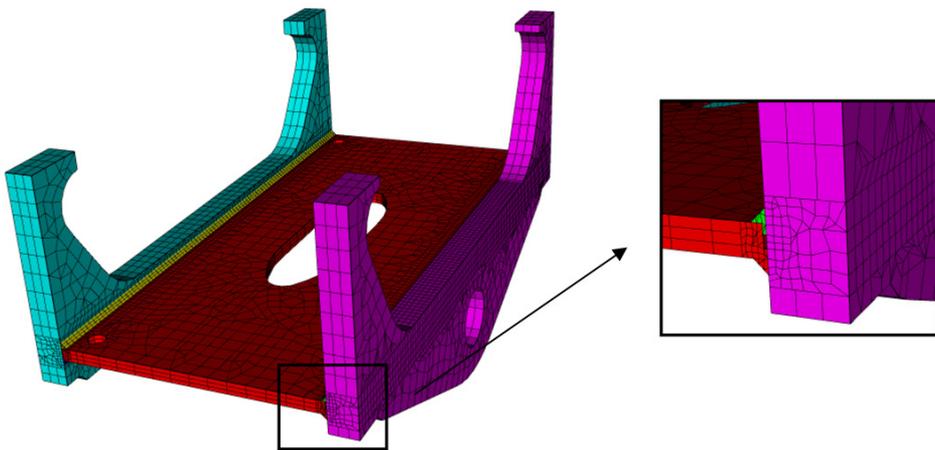
Mesh #	Volume elements	Elements in weld joint (cross-section)
1	13237	5
2	27142	7–9
3	42665	12–15

For each of them, seven welding simulations were run, with an increasing number of block dumps (4, 8, 15, 20, 30, 40, 50). Ten cooling time steps were used in each block dump simulation. The “Fast Simulation” option in VrWeld uses the same number of block dumps for all welds. Therefore, a full-length weld will be welded with the same number of block dumps as a half-length weld. Thus, a welding sequence with full-length welds and, for example, 20 block dumps will be simulated in 90 time steps (4 welds × 20 block dumps + 10 cooling time steps), while a welding sequence with half-length welds will be simulated in 170 time steps (8 welds × 20 block dumps + 10 cooling time steps). The case used to calibrate the mesh and number of block dumps has four half-length welds and two full-length welds (see [Appendix](#)).

The absolute displacement of points P1 and P2 (see [Figure 7](#)) was used to evaluate the results. [Figure 9](#) shows results for point P1, similar results were obtained for point P2. From [Figure 9](#) it can be concluded that there is little difference in results between mesh 2 and 3, while the results from mesh 1 deviate significantly. The difference between mesh 2 and 3 is only 0.1 mm when 40 block dumps are used. Based on these results, mesh number 2 and 40 block dumps will be used for the DSE. The chosen simulation



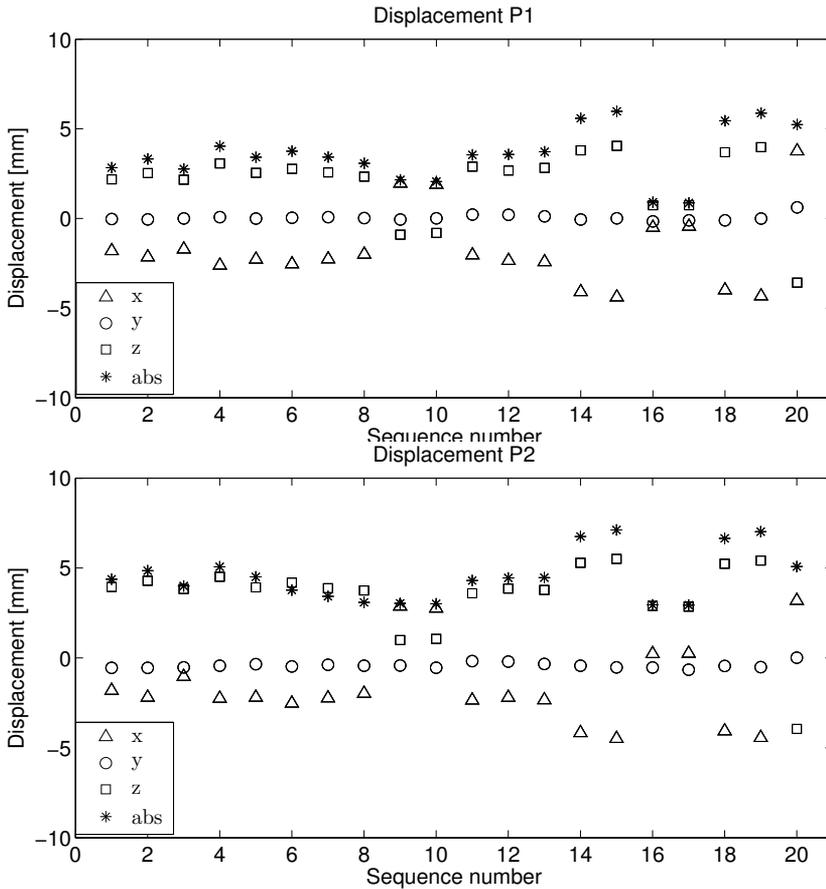
**Figure 9.** Mesh and block dump calibration. There is a difference of 0.1 mm between Mesh 2 and Mesh 3 when 40 block dumps are used.



**Figure 10.** The chosen simulation mesh has 8 elements across the weld joint and 27142 elements in total.

mesh is shown in [Figure 10](#). To verify predictions made by use of the chosen mesh and number of block dumps, the results should be compared with the results from simulations with a moving heat source. If the same best welding sequence is being predicted by the two models, the chosen mesh and number of block dumps are considered to be sufficient.

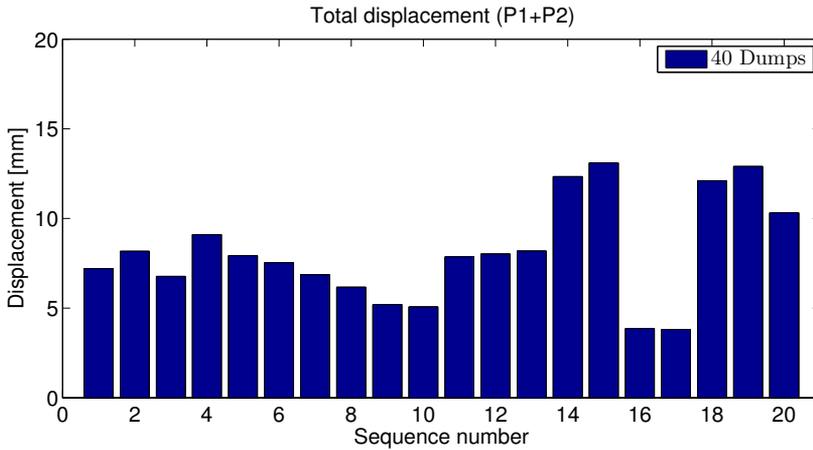
**3.3.5. Design space exploration.** A structure that has four welds, such as the axle bridge, can be welded in 384 ( $2^4 \times 4!$ ) different ways. The large number of combinations is possible when allowing the welds to be laid in any order ( $4!$ ) and in both directions ( $2^4$ ). If only half welds are allowed, the number of possible welding combinations grows to about 10 million possible combinations ( $2^8 \times 8!$ ). Design space exploration (DSE) refers to exploring the set of possible welding combinations and selecting a subset which meets the requirements. In this case, it is not practical to explore the entire design space. To show how welding sequences influence the displacements in P1 and P2, twenty different sequences are



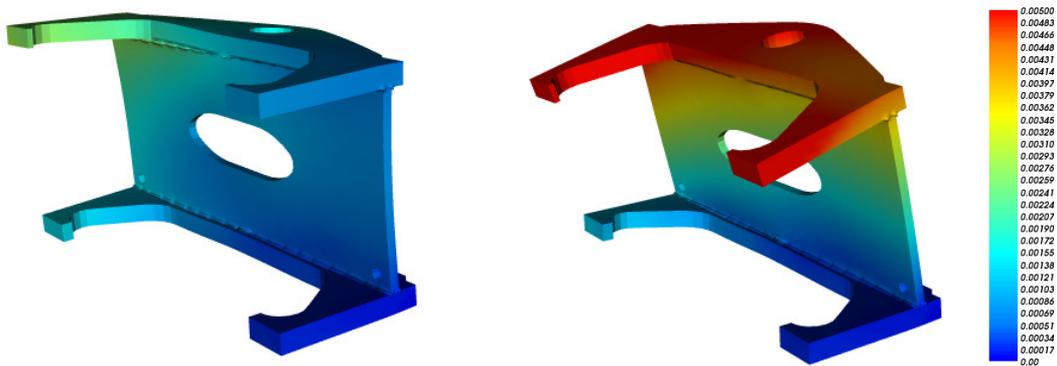
**Figure 11.** Comparison of displacement of points P1 (top) and P2 (bottom) for welding sequences 1–20. The displacement is plotted as x-, y- and z-components and absolute displacement.

simulated; see the [Appendix](#). The results from the twenty welding sequences are shown in [Figure 11](#). Here, the displacement in the x-, y- and z-directions as well as the absolute value is presented for points P1 and P2 for each welding sequence. These results show that welding sequences 9, 10, 16 and 17 give the smallest displacements for P1 and P2, while welding sequences 14, 15, 18 and 19 give the largest displacements in these points. To further illustrate the difference between the welding sequences, the sum of the absolute displacement of points P1 and P2 is shown in [Figure 12](#). This figure shows that welding sequences 16 and 17 give the smallest total displacement (3.8 mm), while welding sequences 15 and 19 give the largest total displacement (13.0 mm). [Figure 13](#) shows the displacement field amplified 20 times for welding sequences 17 and 15.

**3.3.6. Verification of simulation strategy.** To ensure that the chosen number of block dumps gives sufficient accuracy, it is suggested that a small number of moving heat source simulations are performed. In this case, to show that high accuracy has been achieved with the block dumping method, simulations with a moving heat source for all twenty welding sequences were conducted. It was found that the block



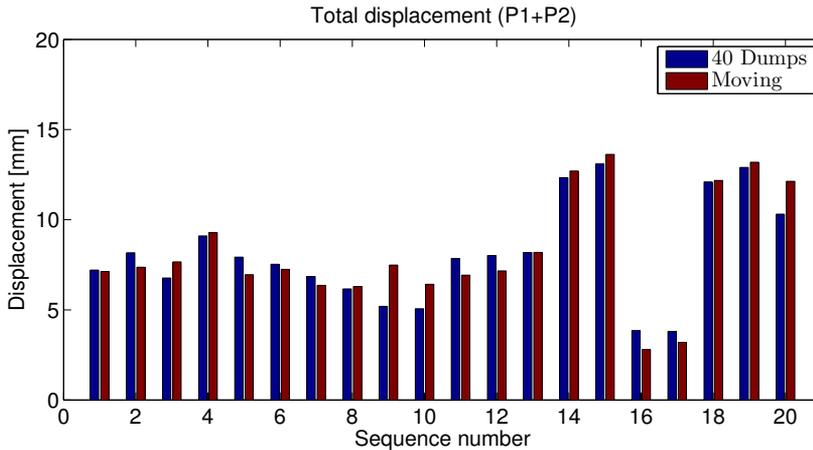
**Figure 12.** Total displacement calculated as the sum of the absolute displacement of points P1 and P2. Notice that welding sequences 16 and 17 give the smallest total displacement (4 mm), while welding sequences 15 and 19 give the largest total displacement (13 mm).



**Figure 13.** Comparison of best (#17) and worst (#15) welding sequences. Colors indicate displacements, given in mm on the scale on the right and magnified 20 times in the geometric representation.

dumped simulations deviate by 0.7 mm (mean value) compared to the moving heat source simulations. Both simulation approaches predict the same welding sequences to give the smallest and largest welding distortions in P1 and P2. Welding sequences 1–4 are welded with full-length welds only and simulated in 170 time steps. Welding sequence 20 is welded with two full-length welds and four half-length welds and simulated in 250 time steps. Welding sequences 5–19 are welded with eight half-length welds and simulated in 330 time steps. The moving heat source is simulated in 450 time steps. Hence, the number of time steps has then been reduced by 60% (welding sequences 1–4), 45% (welding sequence 20) and 25% (welding sequences 5–19).

**3.4. Welding case discussion.** This case study has shown that calibrated block dumping simulations can reduce the calculation time compared to moving heat source simulations. The results from the block



**Figure 14.** Comparison between block dumped and moving heat source simulations.

dumped simulations were able to point out the best (16 and 17) and worst (15 and 19) welding sequences (Figure 14), which was identical to the results obtained with the moving heat source simulations. The block dump calibration and mesh calibration for this case study was performed by running 21 simulations. Thus, using the calibrated block dump method to simulate as few as twenty welding sequences will not save much time. The strength of this method becomes obvious when a larger design space is explored, when hundreds or thousands of welding simulations are performed.

This work is done on the assumption that the 3D transient welding simulations approach used in this paper is able to predict residual stresses and welding distortions correctly. Future work should involve validation of the simulation results obtained in this paper.

This case study has shown that it should be possible to use a simulation-driven design approach in the design process regarding welded structures. Twenty different welding sequences were compared, showing that welding sequences 16 and 17 gave the smallest welding distortions in P1 and P2 and could therefore be considered as the best suited for this structure.

Future work should explore the possibilities of combining the calibrated block dumping simulations with design of experiments or optimization routines to further rationalize the design process of welded structures. A computer program made to generate suitable welding sequences to be explored would significantly decrease designer manual labor. The welding sequences listed in Table 2 were defined manually, which was a quite time consuming task.

It would also be valuable to perform a similar case study where the block dumps are defined by their length instead of the number of block dumps per weld.

#### 4. Simulation-driven rotor dynamical design

Few CAE systems support rotor dynamical analysis, and if they do they have restricted functionality. Therefore, specialized software is commonly used when designing rotating machinery. These specialized rotor dynamical codes often include a lot of functionality, but for nonexperts, they can be difficult to use. In order to facilitate simulation-driven design, an in-house rigid body rotor dynamical demonstrator code

called RESORS has been developed at Luleå University of Technology, Division of Computer Aided Design. The aim is to enable nonexperts to conduct advanced nonlinear rotor dynamical analyses very early in the product development process. By this strategy, sending analysis requests to experts can be avoided, allowing possibilities for exploring a larger part of the design space.

**4.1. Rotor dynamical theory.** Rotor dynamical theory was basically developed during the last century, but started earlier, with Rankine [1869], who incorrectly predicted that the first critical speed could not be exceeded. In the late nineteenth century, de Laval practically proved (see [Childs 1993]) that rotating machinery can run supercritically, which was later verified by Jeffcott [1919]. Since it was introduced, the Jeffcott rotor model has been widely used for different scientific purposes; see for example [Childs 1982; Karlberg and Aidanpää 2003; Karlberg and Aidanpää 2004]. In linear rotor dynamics, eigenfrequencies depend on the spin speed due to the gyroscopic effect. In addition, the direction of the vibration is important, and forward whirl therefore has to be distinguished from backward whirl [Genta 1999]. Some rotor dynamical systems become strongly nonlinear due to for example clearances [Chu and Zhang 1997; Edwards et al. 1999; Ganesan 1996; Goldman and Muszynska 1995; Muszynska and Goldman 1995], processes [Karlberg and Aidanpää 2007], bearings [Harris 1991] etc. Hence, in rotor dynamical software, it is important to enable predictions of linear as well as nonlinear analysis of rotating machinery

**4.2. Simulation strategy.** To show how RESORS rationalizes a design process at early stages of the product development process, an SDD approach is applied on an industrial development case. The SDD approach can (in this example) be divided into three steps, starting with *gathering of information* — requirements, limitations, parameter ranges, etc. Even if nonlinear systems are to be designed, the designer should start with a *linearized simulation model*. This model is primarily used to decide the mesh density and to set suitable damping but also to get an understanding of critical speeds and mode shapes (which may be of interest even in nonlinear systems). To avoid vibration problems due to nonlinearities, at the final step, a *fully nonlinear simulation model* is studied. Since common postprocessing methods fail in nonlinear systems (Campbell diagrams, critical speeds, mode shapes, etc.), other methods are required.

**4.3. Rotor dynamical case study.** In every product development project there are limits to the design space. Before starting modeling and simulation work, information regarding design space boundaries needs to be gathered.

**4.3.1. Information gathering.** In the industrial case used for this paper an overhung rotor system will be designed. The shaft will be supported by radial bearings in order to enable axial displacement during operation. The machinery will be powered by an electrical motor running at 1500 rpm. Earlier concepts were subcritical, but *in the scenario presented here a supercritical machine is to be developed*.

In previous designs, bearings without clearance were used. Table 1, on the next page, lists the needed properties from the latest design, which will be used as a starting point.

The damping ratios are between 0.6% and 16% when considering all vibration modes. In this scenario it is assumed that all parameters shown in Table 1 are fixed except for the rotor mass, the pedestal stiffness, the clearance and the allowed rotor mass. The rotor mass is allowed to be varied between 1800 kg and 2200 kg, the pedestal stiffness between  $10^7$ – $10^9$  N/m and the clearance between 0.01–0.1 mm.

<b>Shaft :</b>	Density	7800 kg/m <sup>3</sup>	<b>Rotor :</b>	Shaft position	Right end
	Young's modulus	200 GPa		Mass	2000 kg
	Length	1 m		Polar mass $I$	200 kg m <sup>2</sup>
	Radius	0.1 m		Transversal mass $I$	100 kg m <sup>2</sup>
<b>Pedestals :</b>	Shaft position	0 / 0.8 m	<b>Load :</b>	Rotor unbalance	10 <sup>-4</sup> m
	Type	Radial, no clearance		Gravity	9.8 m/s <sup>2</sup>
	Isotropic stiffness	10 <sup>8</sup> N/m (both)			

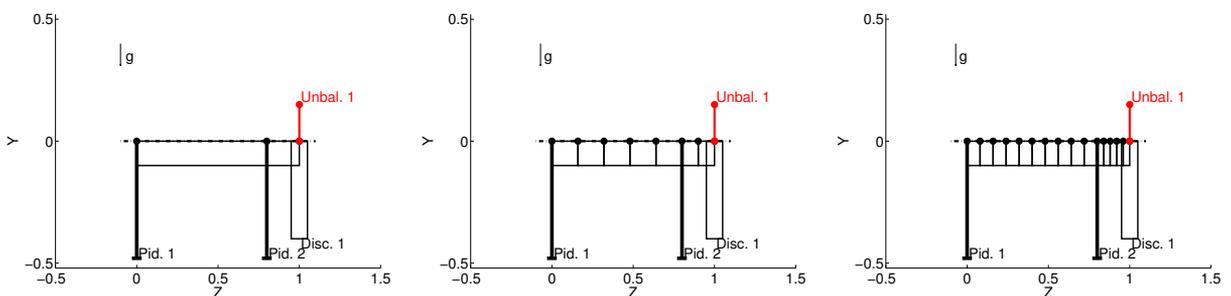
**Table 1.** Properties of latest design ( $I$  = moment of inertia).

**4.3.2. Analysis by use of linearized simulation model.** To rapidly converge on suitable solutions a linearized simulation model, without clearance, is initially used. The objective of this analysis is to find suitable pedestal stiffness and rotor mass that give a supercritical system.

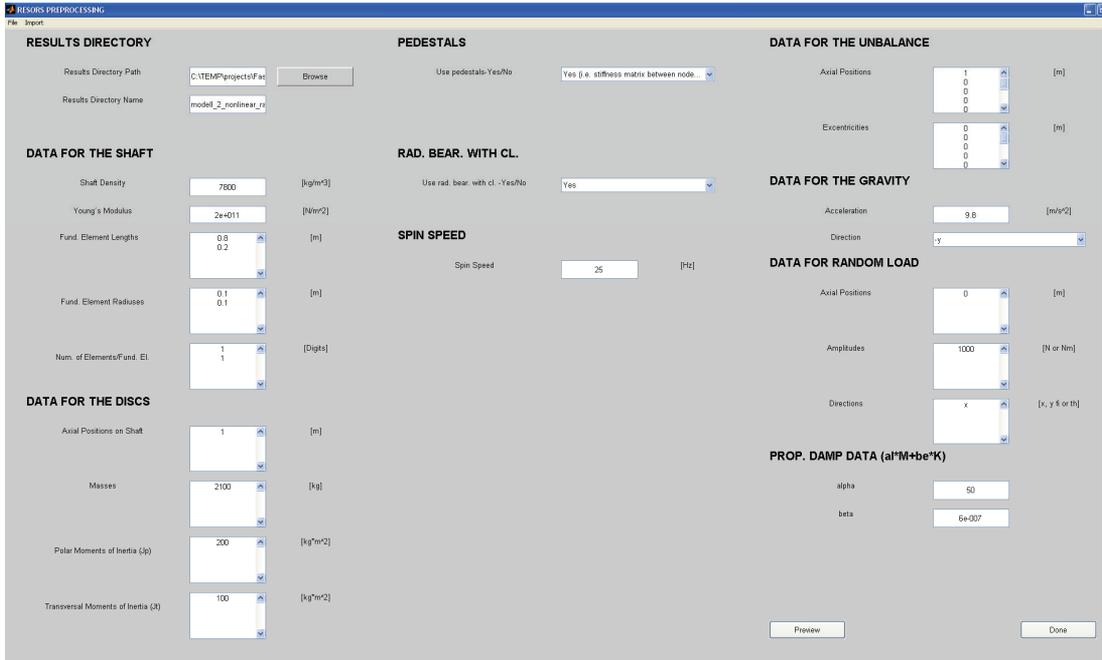
*Preprocessing.* Figure 16a (next page) shows the graphical user interface (GUI) for the preprocessing step in RESORS. Here, the designer is requested to enter data for the model (concept) to be analyzed. The GUI is designed so that the in-data fields are directly coupled to physical properties, and thus easy to understand. The designer enters data about the shaft, the rotor (disc), the pedestals (supporting structure), the load, the spin speed and damping. To simplify implementation of pedestal data, different types of supporting structures (stiffness matrices) are predefined for the designer to test (see Figure 16b). Another common issue to deal with is damping. In RESORS (and other commercial software) proportional damping is implemented and hence two parameters,  $\alpha$  and  $\beta$ , can be chosen to get suitable damping. The choice of  $\alpha$  and  $\beta$  usually requires some experience; therefore, a procedure has been implemented in RESORS showing the minimum and maximum damping ratio for each set of parameters. This gives direct feedback to the designer, who can then choose suitable ranges for the damping ratios (see Figure 16c). Entering all in-data to the preprocessor typically takes a few minutes.

*Mesh convergence.* In the next step of the design process it is suggested that the mesh density is checked by means of variation of eigenvalues. Figure 15 shows the preprocessed model (by use of in-data from Table 1) with three different mesh densities: two, seven and fifteen elements.

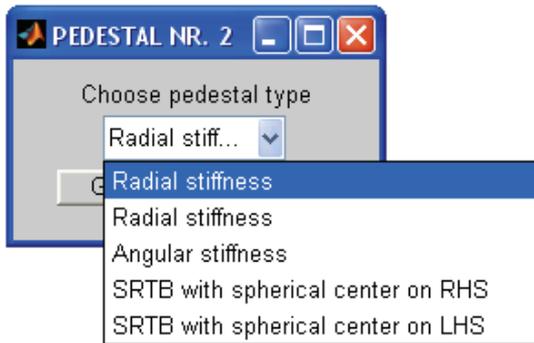
Eigenfrequencies are derived without numerical simulation in time domain. In RESORS, the derivation of Campbell diagrams is therefore implemented directly in the postprocessor and is usually obtained



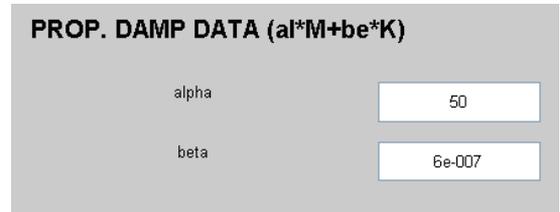
**Figure 15.** Three different mesh densities.



(a)



(b)



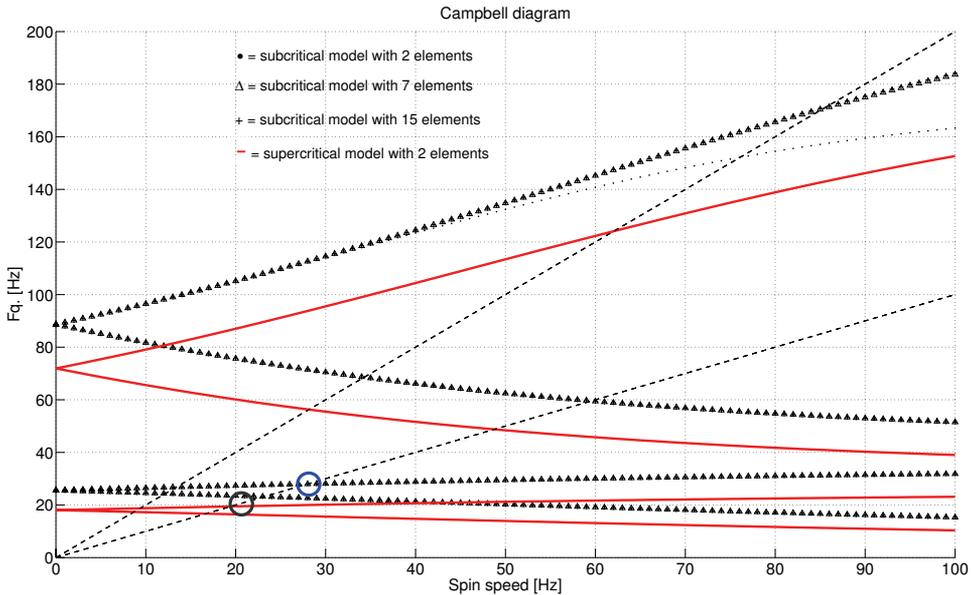
(c)

**Figure 16.** (a) Preprocessing GUI in RESORS. (b) Pedestal selection. (c) Damping ratio selection.

within a few seconds. Figure 17 (page 293) shows how the eigenfrequencies as a function of spin speed (Campbell diagram) depend on the mesh density of the case studied (the black dots, triangles and plus markers). From this figure it can be concluded that, except for the second forward whirl mode at large spin speeds, the eigenfrequencies show little mesh density dependency.

In this scenario, a machine running just above the first forward whirl frequency is to be designed and, hence, even the coarsest mesh gives enough accuracy and will therefore be used henceforth.

*Selection of pedestal stiffness and rotor mass.* When a suitable mesh density has been obtained, the designer can test different concepts by changing parameters in the preprocessor, conduct numerical simulation (when needed) and postprocess and analyze the results. The blue ring in Figure 17 shows that the

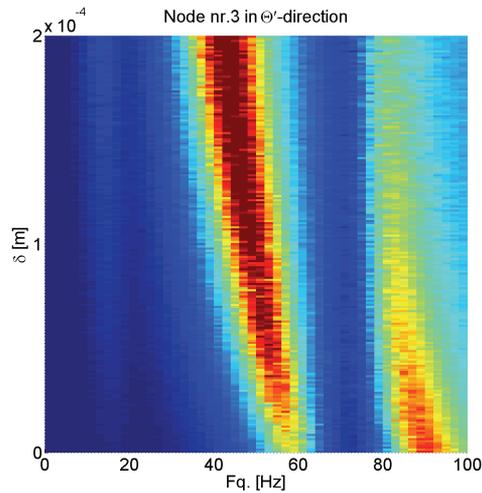


**Figure 17.** Campbell diagram for different mesh densities.

first critical speed, i.e., the intersection between the unbalance excitation frequency and the first forward whirl frequency (dashed line and black dotted line in Figure 17), is somewhere between 25 Hz and 30 Hz. Since a supercritical machine will be designed, the critical speed must be decreased below 25 Hz. The red lines in Figure 17 show the Campbell diagram when increasing the mass to 2100 kg and decreasing the pedestal stiffness to  $5 \times 10^7$  N/m. The gray ring in Figure 17 shows that, by these design justifications, the critical speed is around 20 Hz and, hence, the machine will run supercritically. A set of parameters that give a suitable Campbell diagram can typically be found in a few minutes.

**4.3.3. Analyses by use of fully nonlinear simulation model.** Karlberg [2010] showed that the pedestal stiffness may depend on the amount of clearance in bearings. Karlberg et al. [2010] showed that this clearance can give rise to a significant decrease in resonance frequencies. It was also shown that due to clearance, the unbalance may excite both forward and backward whirl motion. Hence, in order to indicate that the machine will run supercritically as predicted by the linearized model, a fully nonlinear analysis including the clearance must be conducted.

*Analysis of clearance dependency.* In this step of the analysis process, the designer returns to the preprocessor and changes the linear pedestals with pedestals including bearings with clearance, which is done from drop-down lists. In RESORS the possibility of varying parameters has been implemented. This is done by choosing the parameter to be varied from a drop-down list, and then entering the parameter ranges and where on the model to store the data. Since the system is nonlinear, regular analysis measures such as eigenfrequencies cannot be used. Therefore, a transfer function approach has been implemented in RESORS, where random load is applied to the model leading to vibrations, which are measured. Figure 18 shows the transfer function of  $\dot{\theta}$  as a function of the bearing clearance at the rotor when a random load was applied in the  $x$ -direction at the rear bearing. Red indicates high vibration amplitudes,



**Figure 18.** Resonance frequencies as a function of bearing clearance.

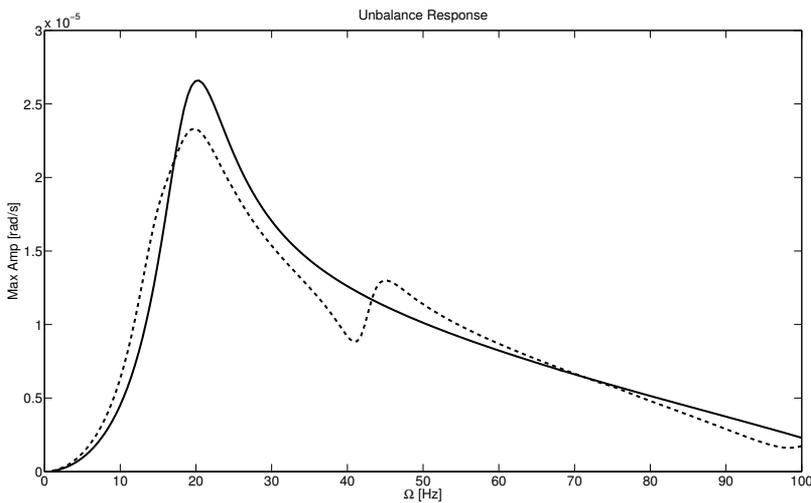
while blue indicates low vibration amplitudes. The model has a spin speed of 25 Hz [Figure 18](#) shows that the first four eigenfrequencies at zero clearance are 16 Hz, 18 Hz, 58 Hz and 92 Hz.

[Figure 18](#) shows that when the clearance in the bearings is increased to 0.2 mm the resonance frequencies decrease up to 20 Hz. The resonance frequencies within this clearance range do not coincide with the spin speed, but a clearance below 0.1 mm is still recommended, since the first multiple of the spin speed (50 Hz) may also excite the system and hence cause resonance. In order to verify that a suitable design has been developed, in the final step the designer runs an unbalance response simulation of the fully nonlinear model. The dashed line in [Figure 19](#) is the unbalance response for the largest acceptable clearance (0.1 mm). As a reference, the unbalance response of the linearized model is also shown (solid line in [Figure 19](#)). Unlike the linear model, analysis of the nonlinear unbalance response shows that critical speeds can occur at more intersections than between the spin speed and the first forward whirl frequency.

The time needed to conduct the nonlinear analysis described above is usually around one hour.

**4.4. Rotor dynamical case discussion.** To show how simulation-driven rotor dynamical design can be conducted a tool named RESORS has been developed and evaluated. The proposed methodology consists of three steps: *gathering of information*, analysis by use of *linearized models*, and by use of *nonlinear models*.

In RESORS, information based on expert knowledge has been implemented, such as different pedestal types, choice of damping parameters, postprocessing tools as transfer functions for nonlinear systems, etc., meaning that the designer does not have to bother about rotor dynamical details. The GUI of RESORS has further been designed to be intuitive and easy to use, featuring e.g., clear descriptions of in-data fields and units, drop-down lists for specific choices, few steps in pre and postprocessing, etc. The post processor in RESORS is designed so that important engineering measures are directly derived without further data processing, e.g., Campbell diagrams, mode shapes, transfer functions and load responses (maximum vibration amplitudes).



**Figure 19.** Unbalance response of linearized model (solid line) and nonlinear model (dashed line).

For the specific case used in this paper an overcritical rotating machine including bearings with clearance was to be developed, meaning that the system becomes strongly nonlinear and, hence, traditional analysis fails. By comparing the results from the linearized model (used to derive a suitable mesh density) with the results from the fully nonlinear model it was found that the resonance frequencies have a strong dependency on the clearance. Hence, a nonlinear analysis is necessary when designing rotating machinery supported on bearings with clearance. Although advanced equations and simulation strategies are required for this type of analysis, with RESORS, analysis is still easy for the designer.

The total time to finish one loop of the proposed three-step methodology is typically around one and a half hours, to be compared with traditional “over the wall” strategies, which usually take days or even weeks per iteration. Hence, by use of the proposed simulation-driven design methodology and effective and intuitive simulation software, a lot of time is saved (days or even weeks). This time can instead be used to test other possible solutions to the problem — in other words, a larger design space can be explored, thereby improving the innovation probability.

## 5. Conclusions

To show how tools for modeling and simulation of mechanics of materials and structures can be designed and used to enable simulation-driven product development two case studies have been used. In both cases, fundamental equations as discussed in Sections 3.1 and 4.1, serve as basis for the software design and development in this paper.

The first case study concerned welding simulations of a Volvo Construction Equipment wheel loader rear axle bridge (mechanics of materials), while the second case study concerned design of a supercritical rotating machine (mechanics of structures).

In the welding case study, a simulation-driven design methodology is applied in order to find suitable welding sequences for a Volvo Construction Equipment wheel loader rear axle bridge. It can be concluded

that by use of a calibrated mesh density and block dumping heat source, the efficiency of the welding simulations was improved with sufficient simulation accuracy.

A program (RESORS) enabling simulation-driven design has been developed and used for the rotor dynamical case study. This software has been designed for improved usability compared to commercial codes used for rotor dynamical analysis through the GUI, being developed purposely for the specific application. The software has been found to be intuitive, efficient and easy to use for designers and improves postprocessing and analysis even for challenging, nonlinear problems.

The methodology used in the two widely different cases both shows that the efficiency in the analysis of challenging problems can be significantly improved through developing advanced tools suitable for use by design engineers. Important features for enabling simulation design by designers are (i) a graphical user interface enabling simulation-driven design; (ii) software built on ambient expert knowledge, made usable for the designer; (iii) significantly decreased time from preprocessing to analysis compared to commercially available codes; and (iv) postprocessing and analysis made easy for designers.

These two examples of simulation-driven design by designers indicate that a larger design space can be explored and that more possible solutions can be evaluated. Therefore, the approach improves the probabilities of innovation and finding optimal solutions.

Furthermore, the calibrated block dumping results presented in [Section 3.3](#) can be used to increase the efficiency of welding simulations when many simulations are required.

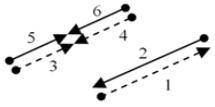
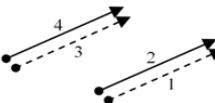
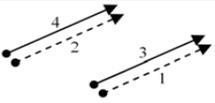
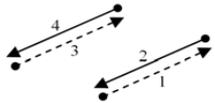
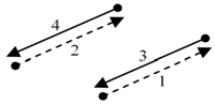
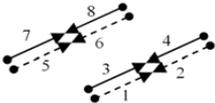
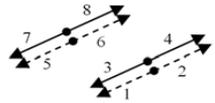
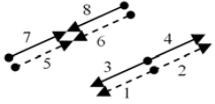
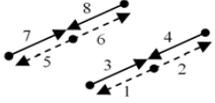
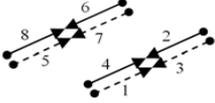
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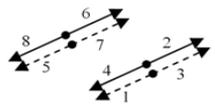
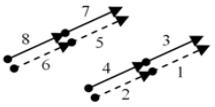
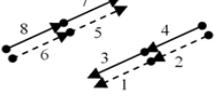
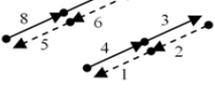
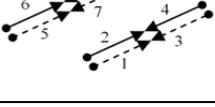
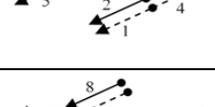
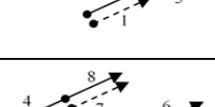
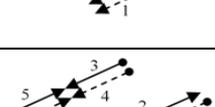
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**Appendix: Design space exploration: simulated welding sequences**

Sequence number	Welding sequence	WeldPath	nbSubPasses	StartWF 1st   2nd	EndWF 1st   2nd	StartTime (s) 1st   2nd
Dump and Mesh Calibration		a (1) b (2) c (3,4) d (5,6)	1 1 2 2	0.0 1.0 0.0   1.0 0.0   1.0	1.0 0.0 0.5   0.5 0.5   0.5	0 158 317   396 475   554
1		a (1) b (2) c (3) d (4)	1 1 1 1	0.0 0.0 0.0 0.0	1.0 1.0 1.0 1.0	0 158 317 475
2		a (1) b (3) c (2) d (4)	1 1 1 1	0.0 0.0 0.0 0.0	1.0 1.0 1.0 1.0	0 317 158 475
3		a (1) b (2) c (3) d (4)	1 1 1 1	0.0 1.0 0.0 1.0	1.0 0.0 1.0 0.0	0 158 317 475
4		a (1) b (3) c (2) d (4)	1 1 1 1	0.0 1.0 0.0 1.0	1.0 0.0 1.0 0.0	0 317 158 475
5		a (1,2) b (3,4) c (5,6) d (7,8)	2 2 2 2	0.0   1.0 0.0   1.0 0.0   1.0 0.0   1.0	0.5   0.5 0.5   0.5 0.5   0.5 0.5   0.5	0   179 158   237 317   396 475   554
6		a (1,2) b (3,4) c (5,6) d (7,8)	2 2 2 2	0.5   0.5 0.5   0.5 0.5   0.5 0.5   0.5	0.0   1.0 0.0   1.0 0.0   1.0 0.0   1.0	0   179 158   237 317   396 475   554
7		a (1,2) b (3,4) c (5,6) d (7,8)	2 2 2 2	0.5   0.5 0.5   0.5 0.0   1.0 0.0   1.0	0.0   1.0 0.0   1.0 0.5   0.5 0.5   0.5	0   179 158   237 317   396 475   554
8		a (1,2) b (3,4) c (5,6) d (7,8)	2 2 2 2	0.5   0.5 0.0   1.0 0.5   0.5 0.0   1.0	0.0   1.0 0.5   0.5 0.0   1.0 0.5   0.5	0   179 158   237 317   396 475   554
9		a (1,3) b (2,4) c (5,7) d (6,8)	2 2 2 2	0.0   1.0 1.0   0.0 0.0   1.0 1.0   0.0	0.5   0.5 0.5   0.5 0.5   0.5 0.5   0.5	0   158 79   237 317   475 396   554

10		a (1,3) b (2,4) c (5,7) d (6,8)	2 2 2 2	0.5 0.5 0.5 0.5 0.5 0.5 0.5 0.5	0.0 1.0 1.0 0.0 0.0 1.0 1.0 0.0	0 158 79 237 317 475 396 554
11		a (1,2) b (3,4) c (5,6) d (7,8)	2 2 2 2	0.5 0.0 0.5 0.0 0.5 0.0 0.5 0.0	1.0 0.5 1.0 0.5 1.0 0.5 1.0 0.5	0 179 158 237 317 396 475 554
12		a (1,2) b (3,4) c (5,6) d (7,8)	2 2 2 2	0.5 1.0 0.5 1.0 0.5 1.0 0.5 1.0	0.0 0.5 0.0 0.5 1.0 0.5 1.0 0.5	0 179 158 237 317 396 475 554
13		a (1,2) b (3,4) c (5,6) d (7,8)	2 2 2 2	0.5 1.0 0.5 0.0 0.5 1.0 0.5 0.0	0.0 0.5 1.0 0.5 0.0 0.5 1.0 0.5	0 179 158 237 317 396 475 554
14		a (1,3) b (2,4) c (5,7) d (6,8)	2 2 2 2	0.0 1.0 0.0 1.0 0.0 1.0 0.0 1.0	0.5 0.5 0.5 0.5 0.5 0.5 0.5 0.5	0 158 79 237 317 475 396 554
15		a (1,3) b (2,4) c (5,7) d (6,8)	2 2 2 2	0.5 0.5 0.5 0.5 0.5 0.5 0.5 0.5	0.0 1.0 0.0 1.0 0.0 1.0 0.0 1.0	0 158 79 237 317 475 396 554
16		a (1,4) b (2,3) c (5,8) d (6,7)	2 2 2 2	0.0 1.0 0.0 1.0 0.0 1.0 0.0 1.0	0.5 0.5 0.5 0.5 0.5 0.5 0.5 0.5	0 237 79 158 317 554 396 475
17		a (1,4) b (2,3) c (5,8) d (6,7)	2 2 2 2	0.5 0.5 0.5 0.5 0.5 0.5 0.5 0.5	0.0 1.0 0.0 1.0 0.0 1.0 0.0 1.0	0 237 79 158 317 554 396 475
18		a (1,5) b (2,6) c (3,7) d (4,8)	2 2 2 2	0.0 1.0 0.0 1.0 0.0 1.0 0.0 1.0	0.5 0.5 0.5 0.5 0.5 0.5 0.5 0.5	0 317 79 396 158 475 237 554
19		a (1,5) b (2,6) c (3,7) d (4,8)	2 2 2 2	0.5 0.5 0.5 0.5 0.5 0.5 0.5 0.5	0.0 1.0 0.0 1.0 0.0 1.0 0.0 1.0	0 317 79 396 158 475 237 554
20		a (1) b (2) c (4,6) d (3,5)	1 1 2 2	1.0 0.0 1.0 0.0 1.0 0.0	0.0 1.0 0.5 0.5 0.5 0.5	0 158 396 554 317 475

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LENNART KARLSSON: [Lennart.Karlsson@ltu.se](mailto:Lennart.Karlsson@ltu.se)

*The Faste Laboratory, Division of Computer Aided Design, Luleå University of Technology, SE-97187 Luleå, Sweden*  
<http://www.ltu.se/tfm/cad/home/d743/d18287/leka?l=en>

ANDREAS PAHKAMAA: [Andreas.Pahkamaa@ltu.se](mailto:Andreas.Pahkamaa@ltu.se)

*Division of Computer Aided Design, Luleå University of Technology, SE-97187 Luleå, Sweden*  
<http://www.ltu.se/tfm/cad/home/d743/d18287/1.52354?l=en>

MAGNUS KARLBERG: [Magnus.Karlberg@ltu.se](mailto:Magnus.Karlberg@ltu.se)

*The Faste Laboratory, Division of Computer Aided Design, Luleå University of Technology, SE-97187 Luleå, Sweden*  
<http://www.ltu.se/tfm/cad/home/d743/d18287/magkar?l=en>

MAGNUS LÖFSTRAND: [Magnus.Lofstrand@ltu.se](mailto:Magnus.Lofstrand@ltu.se)

*The Faste Laboratory, Division of Computer Aided Design, Luleå University of Technology, SE-97187 Luleå, Sweden*  
<http://www.ltu.se/tfm/cad/home/d743/d18287/maglof?l=en>

JOHN GOLDAK: [jgoldak@mrco2.carleton.ca](mailto:jgoldak@mrco2.carleton.ca)

*Department of Mechanical and Aerospace Engineering, Carleton University, Ottawa, ON K1S 5B6, Canada*

JONAS PAVASSON: [Jonas.Pavasson@ltu.se](mailto:Jonas.Pavasson@ltu.se)

*The Faste Laboratory, Division of Computer Aided Design, Luleå University of Technology, SE-97187 Luleå, Sweden*  
<http://www.ltu.se/tfm/cad/home/d743/d18287/jonpav?l=en>